



Section 9

ALI Lessons Learned

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Topics of Discussion



- ♦ Programmatic Issues
- ◆ Technical Issues
- Summary





Programmatic Issues





Programmatic (1 of 3)



- ◆ Issue: The Grating Imaging Spectrometer (GIS) was added to ALI, three months after program start.
 - The GIS integration complexity and cost and schedule impact were underestimated and ultimately led to the elimination of both the GIS and the Wedge Imaging Spectrometer (WIS).
- ◆ Lesson: Mission changes should be considered very carefully





Programmatic (2 of 3)



- Issue: The spacecraft structure was changed from composite to aluminum ten (10) months after program start (February 26, 1997) requiring substantial redesign of ALI structures.
 - The launch vehicle was changed from the Taurus to Delta 2 (January 28, 1997) increasing mass allocations.
 - The Project Office traded spacecraft mass for cost and decided to switch from composite to aluminum structure.
 - The impact of the resulting ALI redesign was not fully appreciated.
- Lesson: Requirements and interfaces need to be frozen early in the design process.





Programmatic (3 of 3)



- ◆ Issue: The decision was made to build all electronics assemblies without Engineering Development Units (EDU) or Qualification Units to "reduce cost".
 - This delayed problem discovery which made correction more difficult and more, not less, expensive.
- Lesson: The EDU serves a useful purpose and should not be skipped.
- ◆ Issue: The mandated 10% reserve proved inadequate for a program for flight-validation of new technologies.
 - The new technologies incorporated in ALI required further development to produce flight hardware.
- Lesson: Adequate reserves should be provided (e.g., 30%), commensurate with technology readiness, to resolve unanticipated problems and to pursue back-up options.





Technical Issues







- Issue: Contamination accumulates over time on the ALI focal plane filters. The contaminants are boiled off through periodic bake-outs on-orbit, every 10 days, lasting 20 hours.
 - While the focal plane is warmed up, the VNIR data are still good, however, the SWIR data are not.
 - Contamination is no longer present in the SWIR bands.
 - Almost all materials used on ALI met NASA outgassing specifications including the Z-306 black paint.
 - All components were baked out. Cost and schedule sometimes drove bake-out duration decisions.
- Lesson: Need uniformity across the board for baking out components to minimize contamination potential in vacuum testing and on orbit.





Leaky Detectors



- Issue: Two ALI detectors (pixels), out of a total of 15,360, are coupling their signal to the neighboring detectors creating streaks in the images.
 - The problem was not evident in the original test data (flood illumination of focal plane detectors).
 - Special algorithms have been developed that greatly reduce the effects of leakage, virtually eliminating it.
- ◆ Lesson: Future Sensor Chip Assemblies should be screened to eliminate any that have leaky detectors.





Vacuum Chamber Window Effects



- ◆ Issue: During optical calibration under thermal-vacuum, it appeared that the focus of the instrument had shifted.
 - It was determined both experimentally and analytically, that the focus shift was due to the chamber window distortion due to the temperature gradient.
 - A technique was developed to eliminate this effect.
- Lesson: Understand all optical effects of thermal-vacuum chamber windows and address them in the test plan and test procedures.





Other Technical Issues (1 of 4)



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- Instrument Alignment on S/C
 - Issue: ALI and Hyperion were not co-aligned.
 - Each instrument's alignment relative to the S/C was carefully measured and recorded.
- Instrument Pointing
 - Issue: It took several weeks to establish accurate instrument pointing on-orbit.
 - S/C pointing was well established.
 - Common Lesson: Need a System Engineer to oversee and correct critical performance issues at the system level.



Other Technical Issues (2 of 4)



Internal Lamps

- Issue: The brightness of the lamps changed on-orbit (increased).
 - Filaments run hotter in zero-G because of the absence of gas convection.
- Lesson: Gas filled lamps are great for checking day-to-day repeatability but should not be used as a radiometric transfer standard.

Subsystem Early Consideration

- Issue: It is difficult to add-on subsystems (e.g., reference lamps) that are not part of the design considerations from the beginning.
- Lesson: Include all subsystems in the early planning.





Other Technical Issues (3 of 4)



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On-orbit Data Processing

- Issue: After launch, the Level 0 data formats changed several times.
 - Shifting of pixels and bands
 - Left-right reversals
- Lesson: The ICD regarding Level 0 processing and the Calibration Pipeline should be completed and frozen before launch.

Schedule

- Issue: The ALI schedule remained very tight even when it became clear that other parts of the program were slipping.
 - Opportunities to avoid overtime and do a more thorough job were missed.
- Lesson: Harmonizing all delivery schedules can produce some program benefits.





Other Technical Issues (4 of 4)



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More Lessons Learned

- Insist on thorough documentation of all vendor (subcontractor) tests.
- Document the "as-built" characteristics of each part.
- Provide a complete photo documentation of the instrument prior to delivery, with close-ups of all critical items.
- Comparison of several independent calibration techniques has proved to be extremely valuable both in ground and on-orbit measurements.
- Calibration of each detector of a large focal plane is a manageable job but requires thorough preparation of test plans, test instrumentation and associated software to process the large volume of data.





Summary



- Many of the lessons learned have a common thread: the tight development schedule and budget require greatly focused mission objectives.
- ◆ A highly motivated, dedicated team can overcome the inevitable problems associated with a high-risk technology validation mission and bring about success.

